



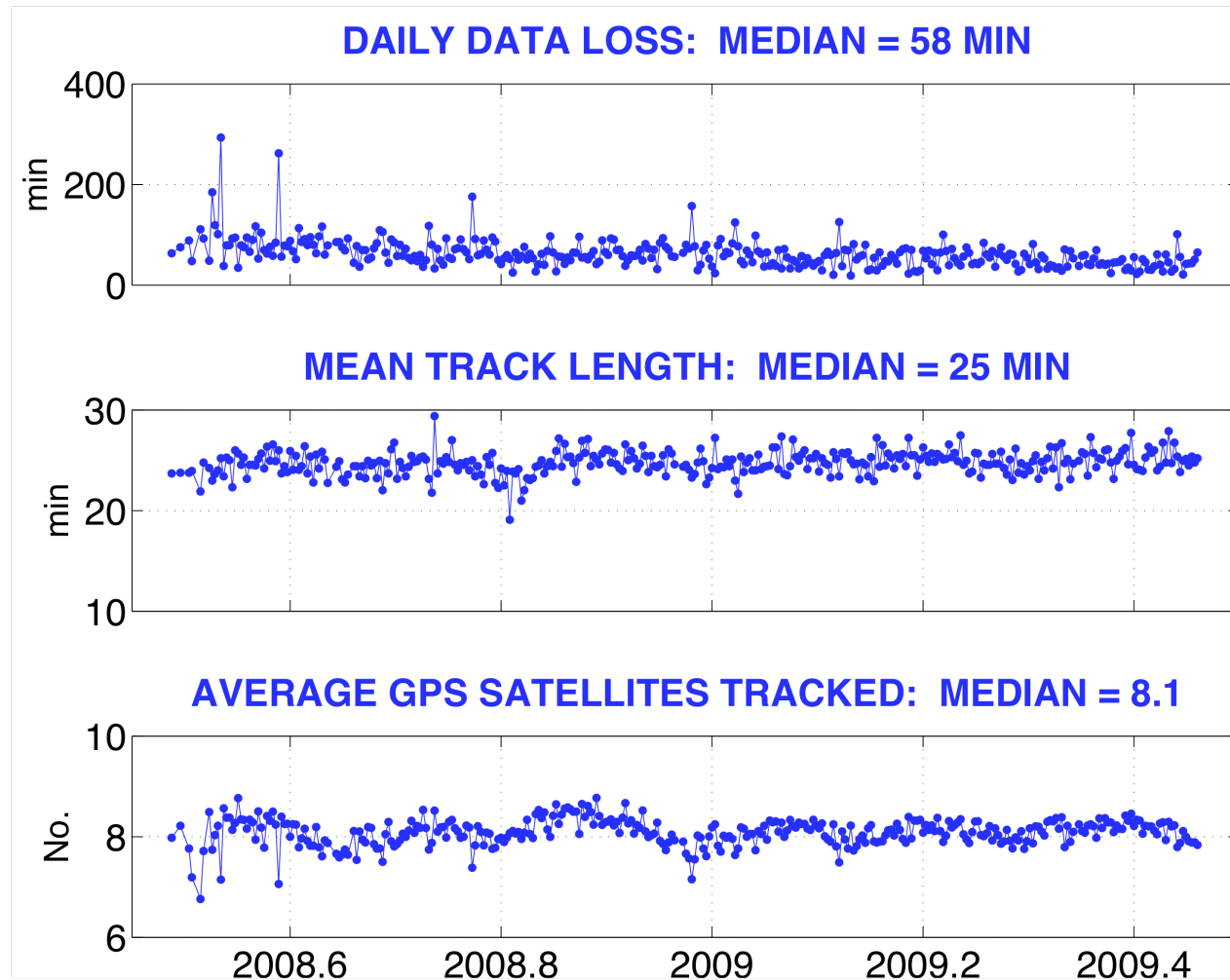
Jason-2/OSTM Precision Orbit Determination with GPS, Model Improvements, Data Processing Improvements

**Willy Bertiger, Shailen Desai, Angie Dorsey, Bruce
Haines, Nate Harvey, Da Kuang, Ant Sibthorpe, and
Jan Weiss**

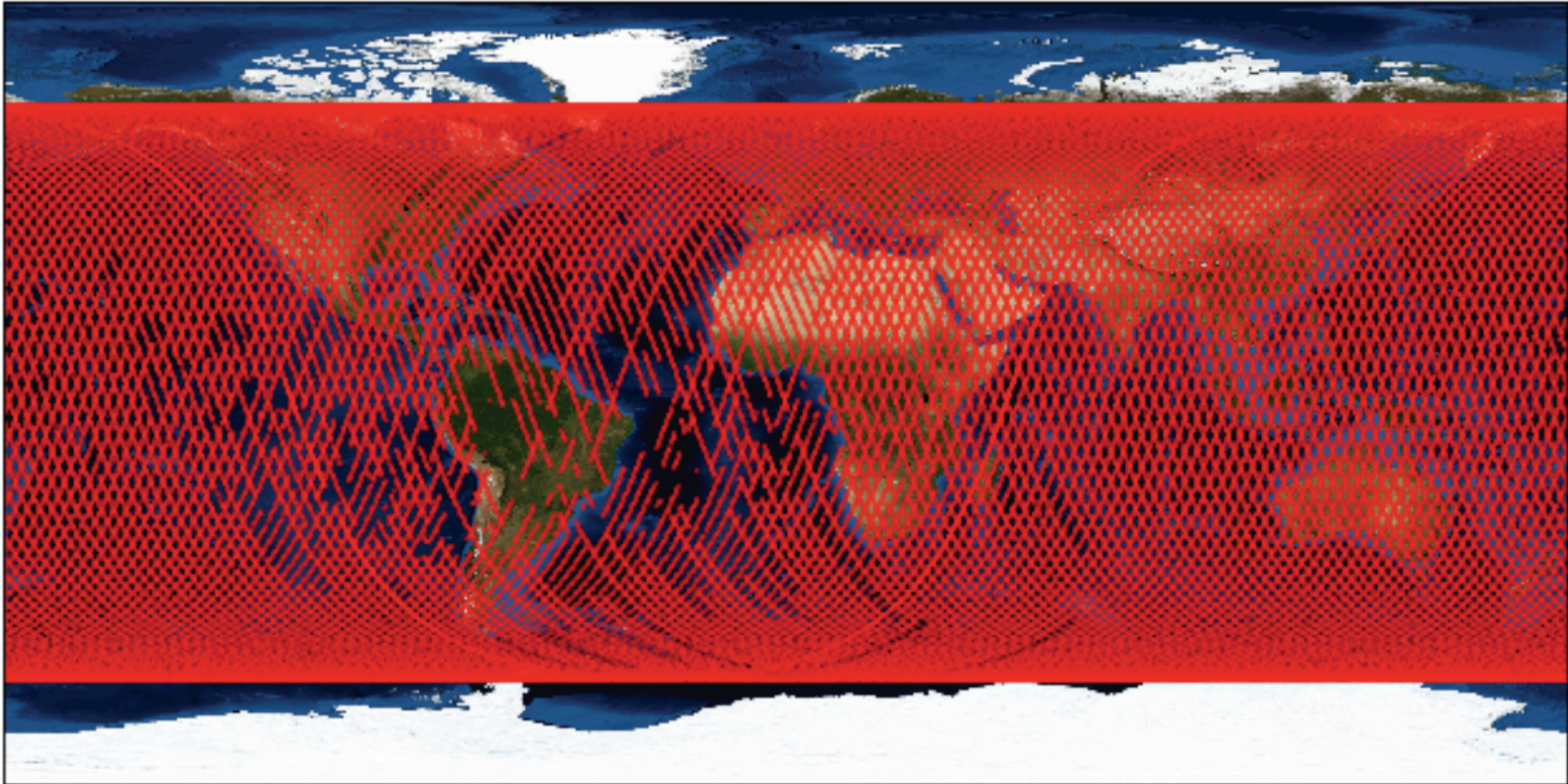
***Jet Propulsion Laboratory, Calif. Inst. of Tech.,
Pasadena CA USA***



OSTM GPS Receiver Daily Tracking Statistics



Typical Data Coverage

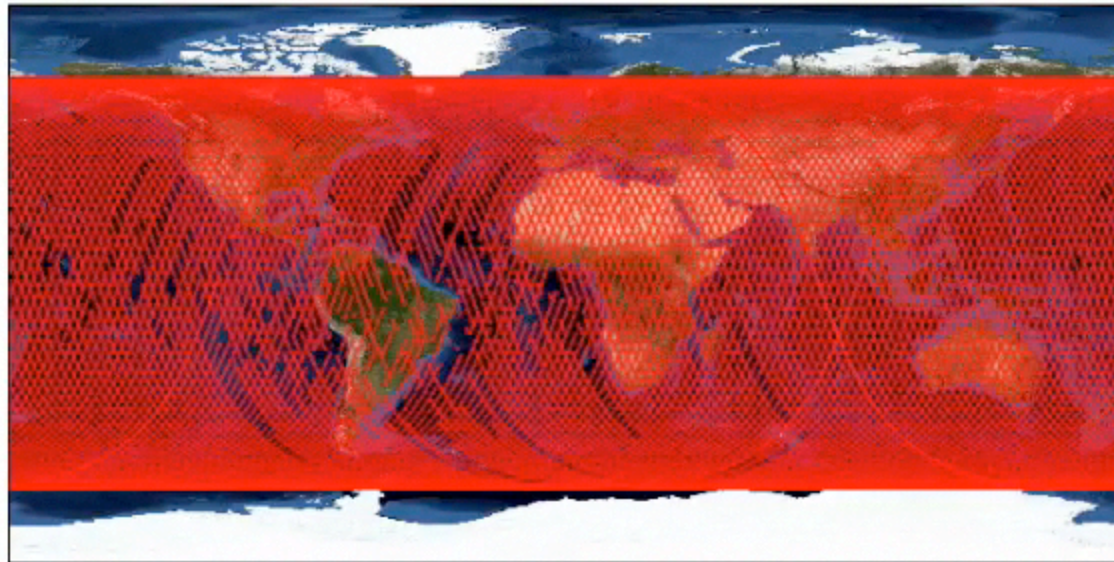


Points on map indicate locations where 4 or more GPS satellites are being tracked for the dates, Aug 10-19, 2008



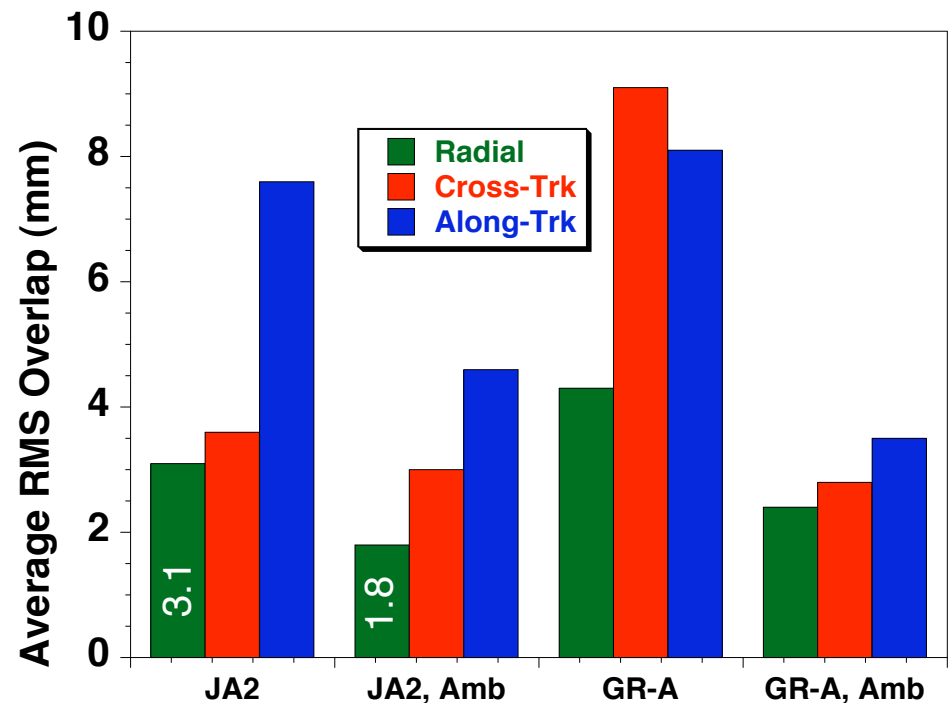
GPS Data Coverage The Movie

Points with ≥ 4 Sats



- **Single Receiver Ambiguity Resolution**
 - Not fixing, finite weight on double-differences
- **Global GPS Orbit and Clock Process (JPL FLINN, QL,...)**
 - For each arc saves – Transmitter name, receiver name, widelane average/standard deviation, phase bias (wlpb file)
- **Single receiver uses orbit/clock and wlpb information and tries to resolve all possible double differences**
 - Widelanes, narrow lanes, iterative improvement
 - Parameter adjustment allows for non-normal error distributions

Overlap Improvement with Ambiguity Resolution



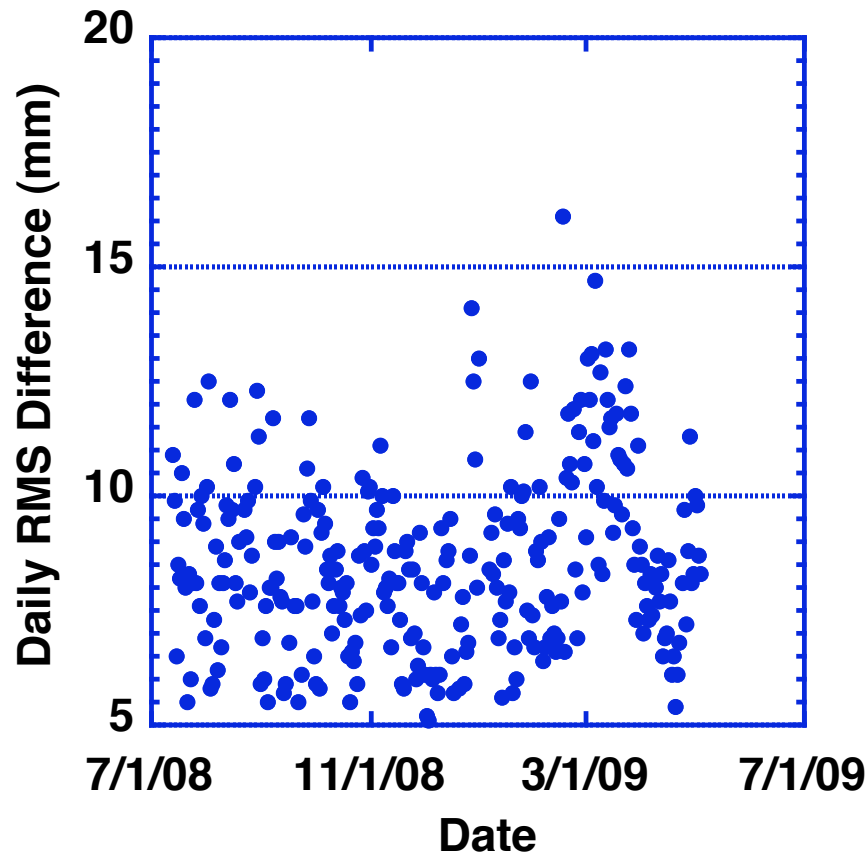


POD Strategy RLSE09A

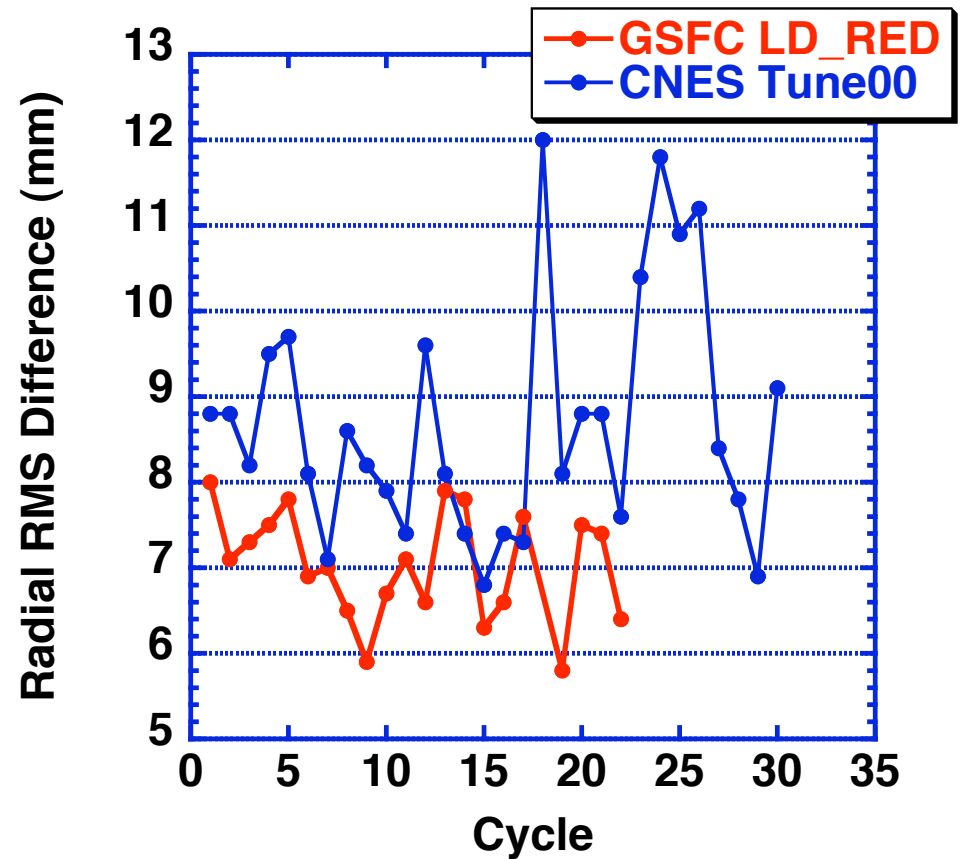


- Ambiguity Resolution
- Reduced dynamics (same as Jason-1 strategy)
- GGM02C (200X200)
 - AOD1B was not used
- Custom Surface force model, Engineering Drawings, Surface Properties
 - Box/Wing + AMR + Altimeter
- New GPS s/c orbit and clock solutions
 - Use IGS phase variation maps
 - Typical 1D RMS overlap (GPS s/c) of 1.5 cm (2.6 cm 3D)
 - Official JPL submission to IGS beginning Sept. 14, 2007.
 - ITRF2005 (using fiducial solution).
- Phase and group delay variation maps for Jason-2 antenna
 - Updated with data from July-Dec. 2008
 - Transmitter reference: IGS offsets, and phase variation maps.
 - Defaults to offset at nadir angles > 14 degrees
 - Receiver reference: pre-launch offset only (no anechoic map)

Radial RMS JPL RLSE09a - CNES Tune00



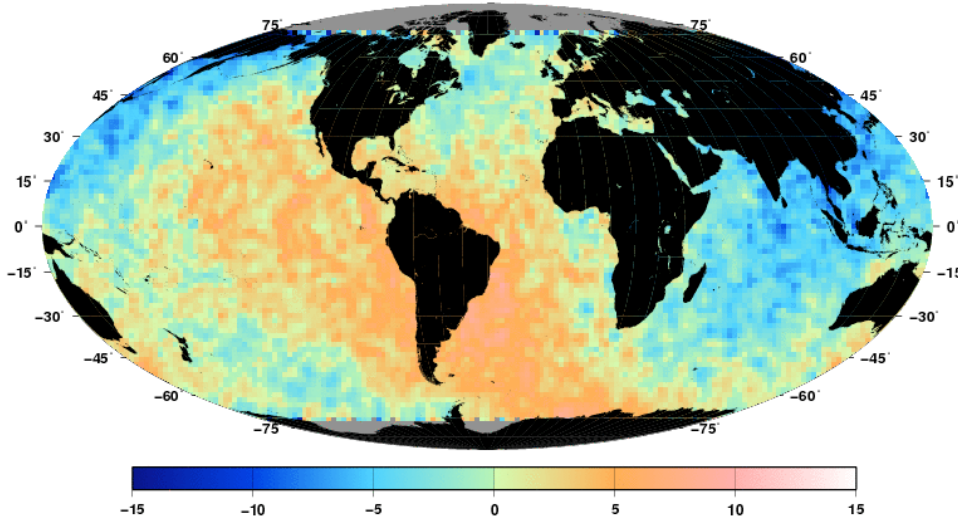
Cycle RMS Difference With JPL RLSE09a



GSFC Mean Cycle RMS = 7.1 mm

CNES Mean Cycle RMS = 8.7 mm

Cy 1–30 Bin RMS = 5.0 mm, CNES Tune00



-15 mm

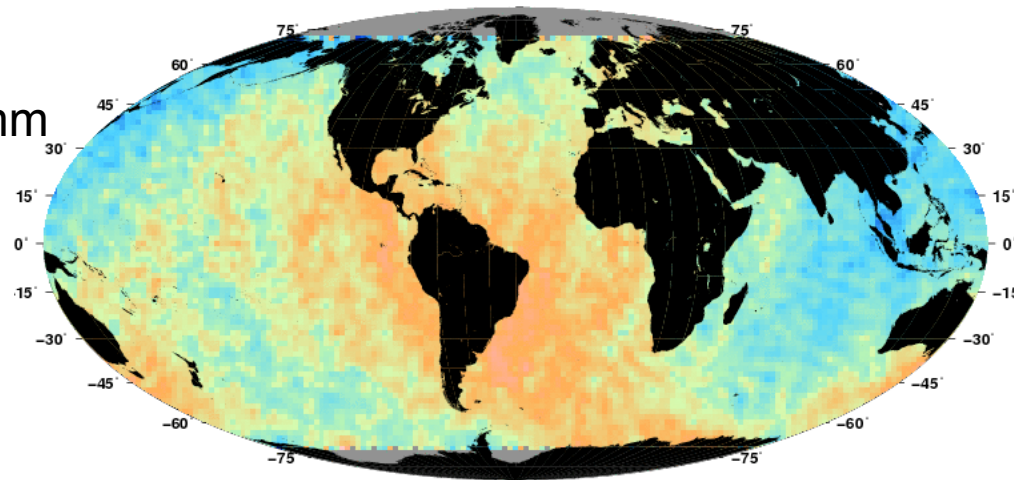
15 mm

Mean XYZ Differences:

CNES – JPL = [-1.1, 3.3, -2.8] mm

GSFC – JPL = [-1.6, 2.1, -1.5] mm

GSFC LD_RED
Cy 1–22 Bin RMS = 4.0 mm

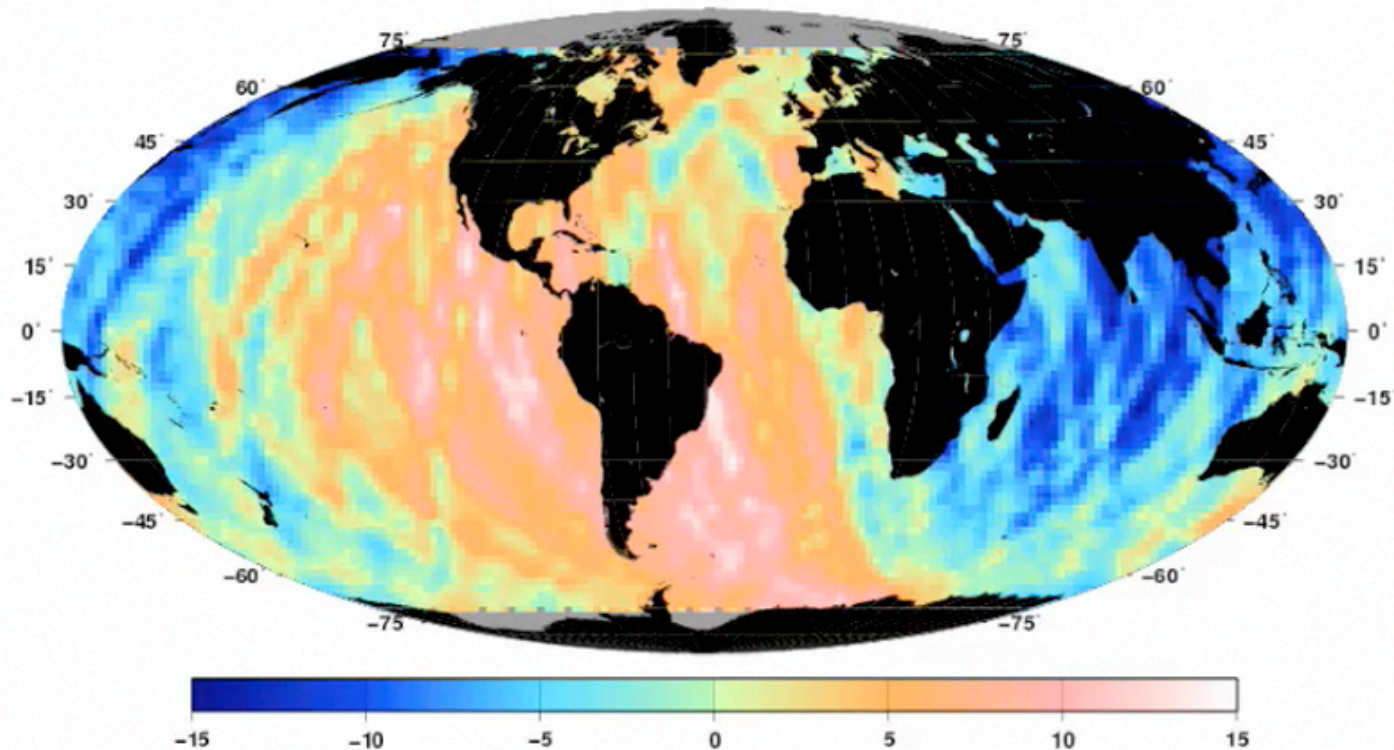


-15 mm

15 mm

CNES – JPL Radial Movie

cycle001





SLR Residual Tests

Pass Mean Statistics, 60 deg. Elev. Cut



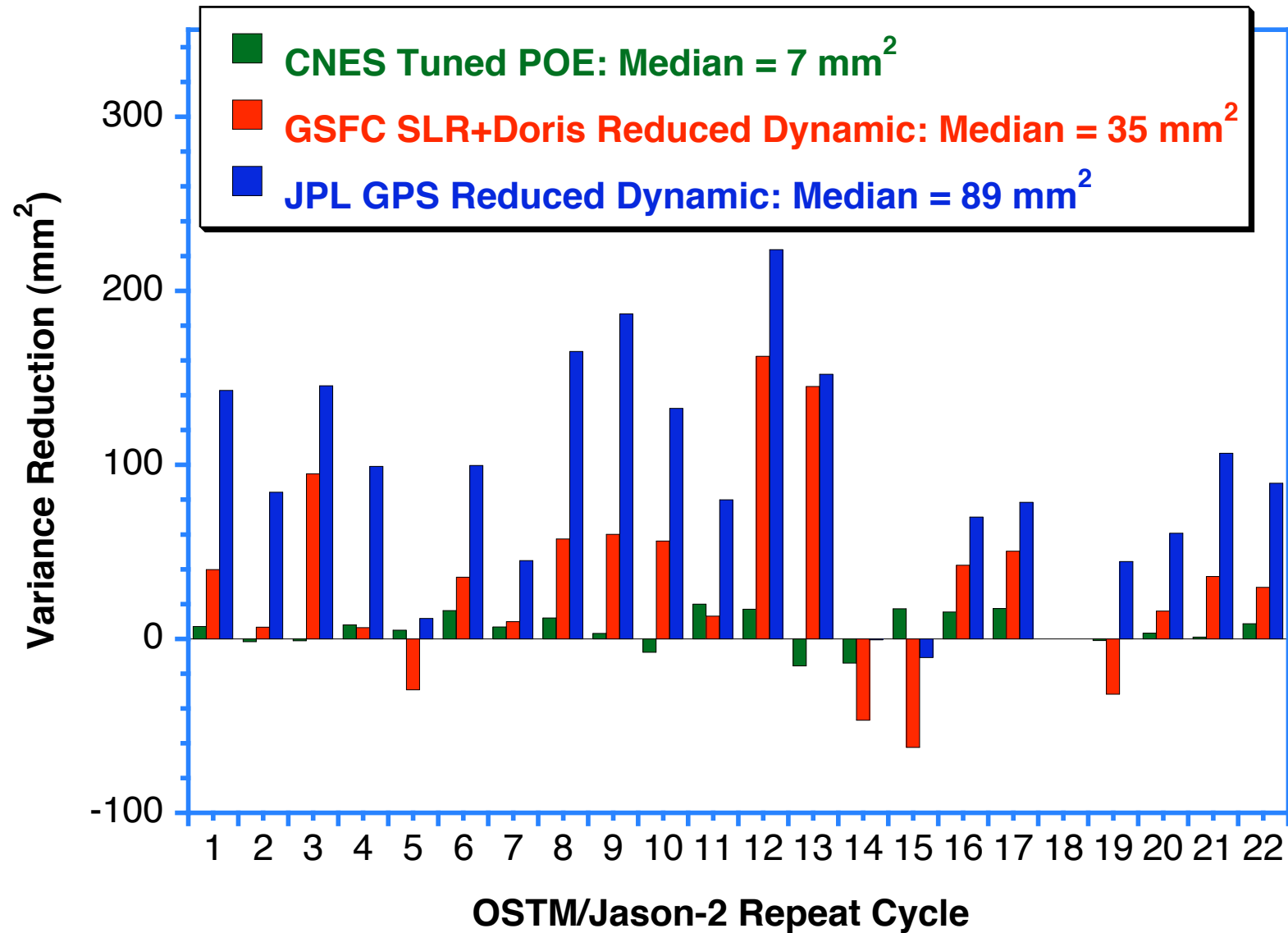
Station	CNES Mean (cm)	JPL Mean (cm)	CNES Std. Dev. (cm)	JPL Std. Dev. (cm)	# Arcs
Monument	1.04	1.08	0.84	0.64	20
Yaragadee	0.28	0.90	0.64	0.62	190
Graz	-0.73	-0.89	0.63	0.80	75
McDonald	1.16	1.05	0.87	0.83	19
ALL	0.13	0.48	0.87	1.04	304
All Weighted			0.67	0.68	

JPL RLSE09a contains no SLR data

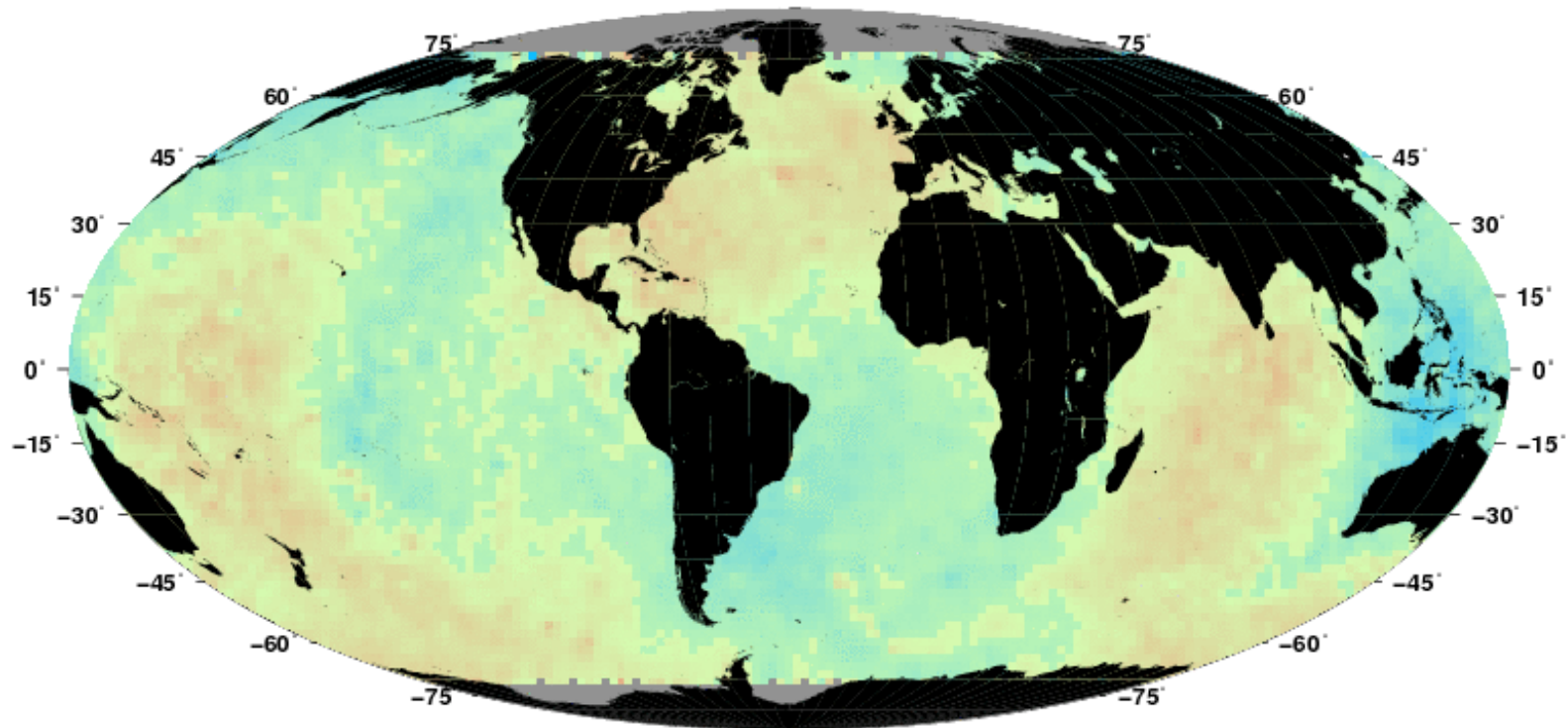


= Larger Std. Dev.

Super Edited Cross-Over Variance Reduction vs. GDR



Cy 1–26 Bin RMS = 0.48 mm, Eigen_tvg – Eigen



-4 mm

+4 mm



Summary Conclusions



- **Better than 5 mm --- Maybe**
 - GSFC Cycle Variance Improvement: 35 mm²
 - JPL RLSE09a Cycle Variance Improvement: 89 mm²
 - GSFC - JPL RLSE09a Typical RMS Difference: 7.1 mm
 - $\text{Sqrt}(7.1^2 - (89 - 35)/2) = 4.8 \text{ mm}$

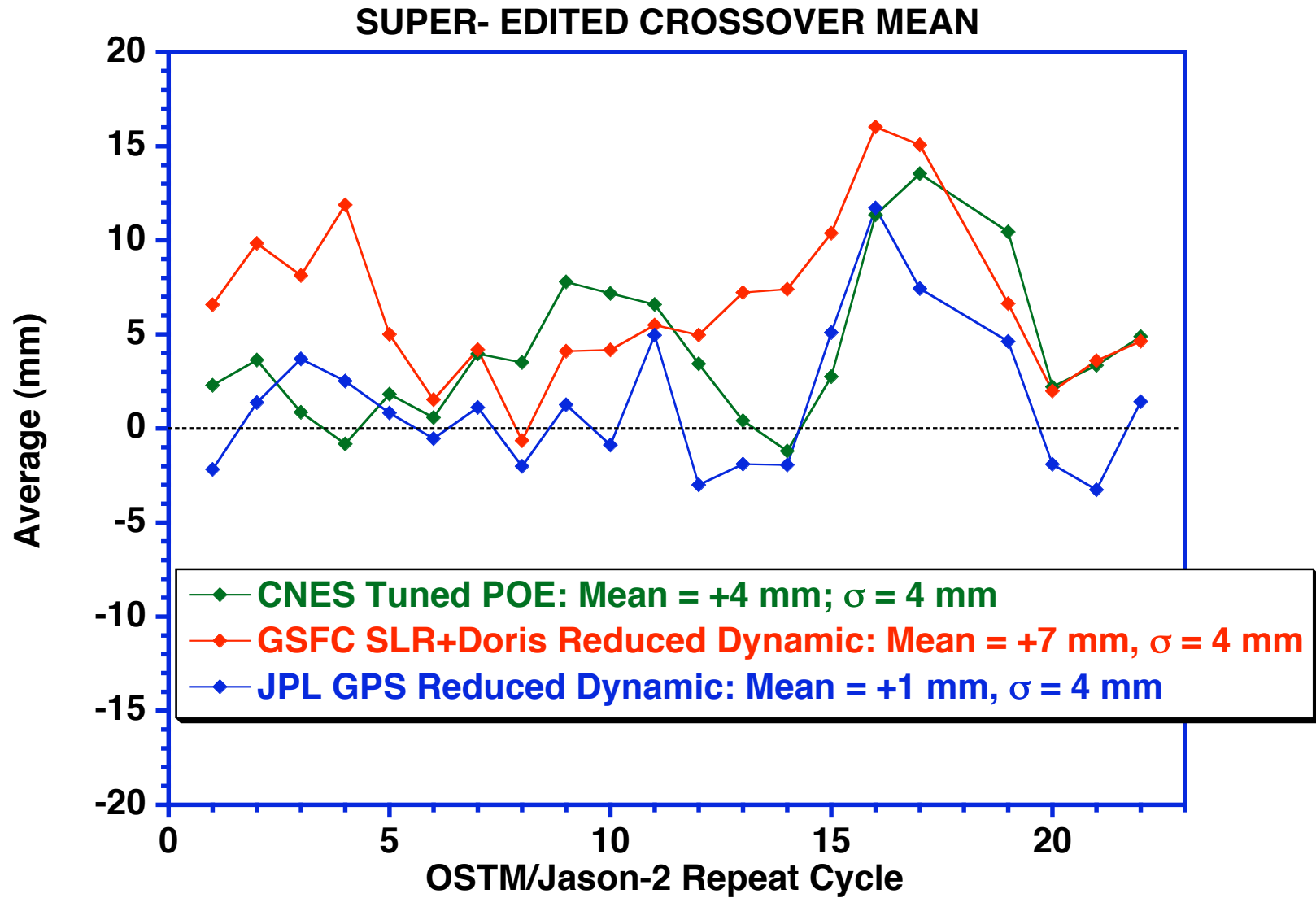
- **Improvements, Future Work**
 - Time Variable Gravity
 - Effects of Data Outage --- Implications for future missions
 - UCL Surface Force Models



Backups Follow



Cross-Over Means



Cy 1–28 Bin RMS = 0.5 mm 09a+AOD–09a

