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## Abstract

In partnership with scientific laboratories and industry, and for several years, CNES has studied the feasibility of a high-resolution ocean topography mission based upon a new class of wide-band Ka-band altimeter. AltiKa payload will be embarked in the SARAL satellite at the same time as the Argos3 instrument, in the frame of a cooperation between CNES and ISRO (Indian Space Research Organization) with an expected launch in 2011. The mission objectives, the instrument description status and pre-flight performances are given in this poster.

## Mission

### AltiKa payload

- > **A Ka-band altimeter**
  - single frequency instrument : ionospheric effects are negligible
  - enhanced bandwidth : vertical resolution and thus error budget improvement
- > **A dual-frequency radiometer (24/37 GHz)**
  - required for tropospheric correction
  - Embedded in the altimeter
- > **A Laser Retro-reflector Array**
  - minimum for orbitography and system calibration
- > **DORIS (Doppler Orbitography and Radiopositioning Integrated by Satellite)**
  - required for achieving very high accuracy orbitography in low earth orbit (orbitography need)
  - measurements link with past or simultaneous missions like T/P, ENVISAT, JASON 1/2 in a well monitored terrestrial reference frame

### Mission objectives

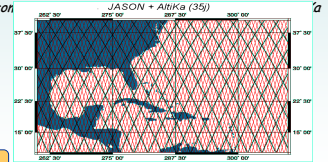
- Central objective :**
- ocean mesoscale circulation
  - data assimilation in a global ocean model
- Contribution to :**
- Continental ice studies
  - Sea-state analysis & forecast
  - Low rain detection and characterization
  - Coastal/inland water altimetry
  - Mean sea level

### Orbit : same as ERS and Envisat

- ✓ Polar (incl. = 98.55°) Sun-Synchronous orbit (6:00/18:00), 800 km altitude, low eccentricity
- ✓ ground track repetitivity period : 35 days

At least 2 satellites are needed to measure mesoscale variations of the ocean

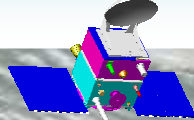
Example of con



### Mission status

- AltiKa payload will fly in the frame of a cooperation with ISRO (Indian Space Research Organisation).
- AltiKa payload will be embarked on the SARAL satellite
- Demonstration of the Ka-band altimetry as a complement to Jason 2 (post-ENVISAT altimetry gap filler)
- Launch foreseen from mid 2011

View of the SARAL satellite



### Rain and cloud effect in Ka-band

#### Effect of rain

- > Rain has an effect on the Signal to Noise ratio
- > CNES/CLS study on rain rates from TRMM/TMI data shows that :
  - Average for one year and all geographical areas show that around 3% of data will be unavailable
  - Unavailability can reach 10% locally depending on season (e.g. Bengal Gof)

#### Effect of clouds

- > Clouds or rain cells smallest than AltiKa antenna footprint introduce distortions on the waveform and errors on parameters
- Study using MODIS data : preliminary results show that this effect increases reasonably the noise on measurements and that for a certain amount of data (less than 10%), a dedicated processing will be necessary

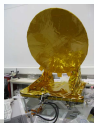
## Instrument description and calibration results

### Main characteristics

Parameter	Value
Altimeter band	35.75 GHz ± 250 MHz
Pulse bandwidth	500 MHz
Pulse duration	110 μs
Altimeter Pulse repetition frequency	~ 3.8 kHz (adjustable along the orbit)
Echo averaging (altimeter)	~ 25 ms
Altimeter Link budget	11 dB (sigma naught = 6.5 dB)
Antenna diameter	1000 mm
Radiometer band	23.8 GHz ± 200 MHz 37 GHz ± 500 MHz
Radiometric resolution	< 0.4 K
Radiometric accuracy	< 3 K
Radiometer averaging	200 ms
Data rate	38 kbits/s
Mass (altimeter+radiometer)	< 42 kg
Power consumption (altimeter+radiometer)	< 100 W

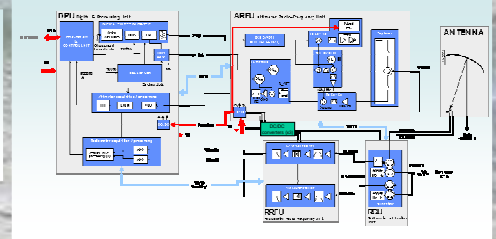
The antenna is common for altimeter and radiometer and is composed of

- o A fixed offset reflector
- o A three-band feed
- o A sky horn pointing to deep space



The AltiKa instrument consists of a Ka-band altimeter which functions are based on proven concepts and already developed sub-systems, as it inherits from Sival (CRYOSAT mission) and Poseidon3 (JASON2 mission), and an embedded dual frequency radiometer.

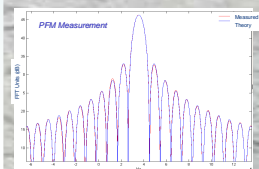
- Altimeter function is based on the full Deramp technique
  - Radiometer function is based on the direct detection principle. The radiometer radiofrequency unit (RFU) consists of two RF receivers, which are developed by EADS ASTRIUM.
- The radiometer must be switched-off during radar altimeter emission.



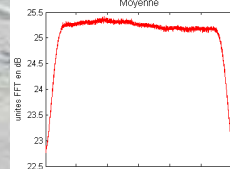
Block diagram of the instrument

### Calibration modes

- Calibration-1 : provides with the altimeter point target response (complex spectrum).
- Calibration-2 : provides with the altimeter receiver transfer function after Deramp (in the frequency domain) by averaging the natural thermal noise in the reception channel over a long period.
- Radiometer calibration : RCU can commute every N seconds to a sky horn pointing to deep space (cold reference) or to a load at ambient temperature (hot reference)



AltiKa measured Point target response (CAL1 mode)



AltiKa CAL2 measurement

- o Cal1 and Cal2 are Compliant to requirement
- o Very good stability of the PTR
  - Versus frequency
  - Versus T°

SARAL Payload Integration Module after thermal vacuum

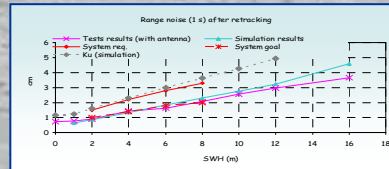
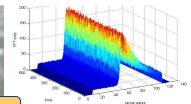


## AltiKa pre-flight measured performances

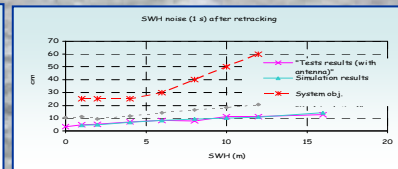
### Range and SWH performance

- > Accuracy of the altimeter range measurement over sea surface : about 1 cm for a SWH of 2 meters
- > Improvement of about 40% on the range noise versus Ku-band performances
  - o Better estimate of the velocity fields (topography gradients) and better analysis of the eddies structure along-track

Test bench simulated waveforms

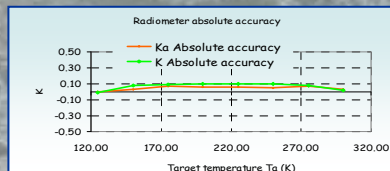
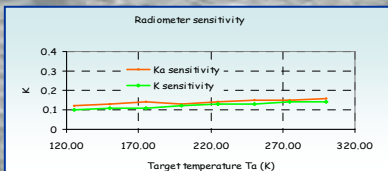


1 second range noise (cm) versus SWH : ground tests results



1 second SWH noise (cm) versus SWH : ground tests results

### Radiometer performances

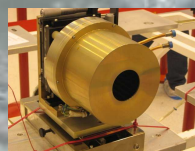


Radiometer performances during acceptance tests, for an ambient temperature of 20°C, without antenna

- Worst case, in temperature and with antenna, absolute accuracy < 1.3 K
- Fully compliant to the specifications
  - o Ka sensitivity req. < 0.4 K
  - o K sensitivity req. < 0.3 K
  - o Absolute accuracy req. < 3 K

- Acceptance tests in thermal vacuum using radiometric target in front of antenna horns.
- Dynamic of brightness temperature from 110 K to 310 K, for 3 ambient temperatures (0°C, 20°C and 40°C)
- Calibration of the radiometric model developed by TAS-F (implemented in ground segment)

⇒ Estimation of the radiometer performances (in terms of sensitivity and absolute accuracy), at 20°C



Black body (radiometric target) Accuracy of 0.25 K



Tests in thermal vacuum involving cryogenic means