Systematic Differences in the Center-of-Origin Realization of Jason-1 and Envisat

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OSTST Annual Meeting, Lisbon, October 19, 2010



Differences in the origin realization of Jason-1 and Envisat

Multi-Mission Crossover Analysis (MMXO)

Envisat Orbit Solutions

Improvements made by reprocessing

Differences between CNES and ESA reprocessing

Differences in the realization of origin for Envisat and Jason-1

Conclusion



MMXO takes advantage of the high redundancy provided by a multiple surveying of the sea surface through contemporaneous altimeter missions.

The redundancy is expressed by short-term single and dual satellite crossover differences Δx_{ij} in all combinations.

Together with consecutive radial errors δx_i they are minimized by a least squares adjustment, which includes a variance component estimate to achieve an objective relative weighting between different missions.

Main steps:

- 1) Computation of single and dual-satellite crossover differences Δx_{ij} in all combinations
- 2) Minimizing both $\Delta x_{ij} = x_i \cdot x_j$ and $\delta x_i = x_{i+1} \cdot x_i$ and estimation of radial errors x_i at all crossover points within a **least squares adjustment**
- 3) Derivation of relative range biases, <u>center-of-origin shifts</u> as well as common error components of ascending and descending passes



Orbit Solutions available for ENVISAT

Originally GDR orbits

- GDR-A (009-040), GDR-B (041-067), GDR-C (068-070), GDR-C' (since 071)
- Inhomogeneous, partly based on GRIM gravity field and ITRF2000

Version	Cycle	Gravity field	Reference	Tracking
			System	Systems
GDR-A	09-40	GRIM5	ITRF2000	DORIS/SLR
GDR-B	41-67	EIGEN-CG03C	ITRF2000	DORIS/SLR
GDR-C	68-	EIGEN-GL04S	ITRF2005	DORIS/SLR

Reprocessed CNES orbits

- available for cycles 015 071, GDR-C' POE standard
- not available for the first part of the mission (about 0.5 years)

Reprocessed ESA orbits (sol6)

• available for whole mission lifetime (cycles 009-090)



Orbit improvements due to Reprocessing



Differences in the origin realization of Jason-1 and Envisat

Orbit improvements due to Reprocessing



Differences in the origin realization of Jason-1 and Envisat

Differences between CNES and ESA reprocessing



- maximal difference of 3 mm
- differences in reprocessing solutions 1 mm
- => Orbit solution do not change the range bias of Envisat



Differences between CNES and ESA reprocessing



Differences in the origin realization of Jason-1 and Envisat

rel. center-of-origin shifts of Envisat w.r.t. Jason-1



Differences in the origin realization of Jason-1 and Envisat

rel. center-of-origin shifts of Envisat



Differences in the origin realization of Jason-1 and Envisat

What is the reason for this trend ???

it is visible for all orbit solutions (CNES/ESA/GSFC) it is probably caused by Envisat may also be a long-period oscillation significant only in the y-component

Possible explanations:

- \Rightarrow Solar Radiation pressure model
- \Rightarrow time variable gravity field (different orbit heights)
- \Rightarrow tracking system differences
- \Rightarrow reasons other than orbit

 \Rightarrow ... ???



What are the consequences of this trend ???

 \Rightarrow time-dependent geographically correlated errors



Differences in the origin realization of Jason-1 and Envisat

relative GCE of Envisat (w.r.t. Jason-1)





Differences in the origin realization of Jason-1 and Envisat

Mean sea level trends

taken from: Faugere et al. (2010): *Envisat ocean altimetry performance assessment*, Living Planet Symposium, Bergen





Differences in the origin realization of Jason-1 and Envisat

✓ MMXO can reveal range bias between different altimeter missions as well as information on the center-of-origin realization and geographically correlated errors. It is independent from orbit configuration (no repeat tracks, formation flights, etc. necessary).

✓ New, consistent Envisat orbits for the whole missions lifetime are available. There are no significant differences between reprocessing solutions from CNES and ESA.

✓ Small but significant differences in the realization of orbit origin of Envisat and Jason exist, mainly in the y-component. The reasons are still unknown.





Missions included in MMXO





Differences in the origin realization of Jason-1 and Envisat

Separation of radial errors into range bias and center-of-origin shifts Least square adjustment for each 10-day cycle

$$x_i + \varepsilon_{x_i} = \Delta r + \Delta x \cos \varphi_i \cos \lambda_i + \Delta y \cos \varphi_i \sin \lambda_i + \Delta z \sin \varphi_i$$

input:radial errors x_i at location φ_i , λ_i output:mean range bias Δr mean center-of-origin shifts Δx , Δy , Δz



Radial Errors are available for ascending and descending tracks.

From the differences between ascending and descending errors (mean values per 2.5° by 2.5° region) the mean GCE can be computed:

 $\Delta \gamma = (dr^{asc} + dr^{desc})/2$ [Rosborough, 1986]

 dr^{asc} average of the radial errors (mean reduced) of all asc. passes dr^{desc} average of the radial errors (mean reduced) of all desc. passes $\Delta\gamma$ mean of ascending and descending errors, GCE per cell

