

Ocean Surface Topography Science Team Meeting Precision Orbit Determination Splinter

Reducing the Dependency of Precise Orbits on Time Varying Gravity Field Modeling Errors

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- Errors in the modeling of secular variations of the gravity field cause signatures which are significant for regional MSL applications
- □ Current approach (GDR D)
 - GDR orbits are DORIS+SLR solutions including GPS when available
 Same approach for Jason-1, Jason-2, Envisat, Cryosat, HY2A
 - Secular variations in gravity field are taken into account by 50x50 drifts included EIGEN-GRGS_RL02bis_MEAN-FIELD field
 - This linear model provides significant improvement in the orbit performance wrt to the previous generation of orbits
 - Altimetry, SLR residuals comparison with reduced-dynamic GPS orbits (Jason)
 - Comparison with orbits obtained using the 10-day time series of GRACE derived fields

Context, Scope and Objectives

The difference between orbits using the mean model (GDRD) and the time series of 10-day fields is quite stable (expected radial accuracy of DORIS-only dynamic orbits is < 1.5 cm RMS)</p>



Context, Scope and Objectives

Stable RMS of radial orbit difference between GDRD DORIS-only dynamic orbits (24hr 1/rev) and JPL reduced dynamic GPS orbits

- Orbit difference increases when drifts are removed
- Stable orbit differences (RMS) between Grace-based TVG orbits and JPL11A



Context, Scope and Objectives

Limitations :

- We cannot safely extrapolate linear trends if the underlying geophysical processes are not linear; we need a solution for POD in case GRACE data were lost
- Several potential contributors to MSL trends estimates (Cryosat-2 and SARAL-AltiKa) rely only on DORIS+SLR data : reduced-dynamic approach is limited by observability
- The performance of GPS-based reduced-dynamic orbits with Sentinel-3 GPS receiver remains to be assessed

Interest in assessing the capability of DORIS to observe local mass variations on the long-term

• **DORIS-only dynamic orbits** (24hr 1/rev , 1 drag coeff every 3 revs)

Sensitivity to variations in spherical harmonics



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Point Mascon Model : a simpler approach

Alternative: solve for the mass of N point mascons at known locations

Acceleration on the satellite induced by the *i-th* point mascon

$$f_i = -\frac{\mu_i}{d_i^3}\vec{d}_i + \frac{\mu_i}{r^3}\vec{r}$$

Additional term ensures total mass conservation

No constraint applied to conserve the center of mass position – verified a posteriori (see backups)





Point Mascon Model : motivation

Even a single Point Mascon (Greenland) is quite representative of the difference in the geopotential acceleration between 2005 and 2012



Cryosat , arc 094 - Point Mascon Fit (Greenland : -40°,72°) - Radial Component

Point Mascon Model : method



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- We use only DORIS data from Jason-1, Jason-2, Cryosat-2, Envisat - Observability is brought by the low-altitude-high-inclination satellites (Envisat + Cryosat span 2002 – 2012 period)
- The normal equations from all satellites are stacked over 180-day intervals to obtain a time series of mass values
- This Point Mascon model is then used in Doris-only dynamic POD : mass values are solved-for on an arc-by-arc basis, but tightly constrained to this empirical model

□ Validation :

- Mascon mass trends compared to GRACE EWH time series
- Comparison to orbits using the 10-day time series of GRACE fields
- Independent SLR residuals on DORIS orbits



DORIS Mascons Vs GRACE time series of EWH







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Envisat: 10-day series – GDRD w/o drifts



Envisat: 10-day series – GDRD



Envisat: 10-day series – GDRD w/o drifts + DORIS mascon model



□Trends in the hemispheric radial differences of Envisat orbits without Grace-derived drifts are reduced from 3 mm/yr to <1 mm/yr when the mascon model is used

Orbit solutions using mascon model without drifts are close to GDRD solutions







Radial orbit difference with respect to JPL11A reduced-dynamic GPS orbits is also slightly reduced when the mascon complement is used



- DORIS measurements from all altimeter satellites provide sufficient observability to solve for a 6-mascon complement to the standard GRACE-derived fields
- This approach improves the orbit solutions over the 2002-2012 time span, especially when GRACE-derived drifts are removed from the gravity model
- ❑ As a result, the mascon empirical approach is promising in maintaining and monitoring the DORIS-based orbit performance in case of loss of the GRACE time series, to the extent that locations where the strongest long-term variations occur remain known over a 3-4 year time-span
- Prospects : include SPOT data (SPOT-2 overlaps with TOPEX), GPS and SLR when available, compare to classic SH approach



Backups



DORIS Mascon model

Center of mass position of the DORIS Mascon model





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