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The effect of geocenter motion on Jason-2 orbits and the mean sea level

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Abstract
We have investigated the impact of geocenter motion on Jason-2 orbits. This was accompliated by computing a series of Jason-2 GPB-based and SLRDORRS-based orbits using TRF2006 and the IG506 framework based on the most recent GSFC 0906 standards. From these only, we stract the Jason-2 orbit frame nanalisional generators pay cycle by the means of a Heijorn transformation between as at of internor contas as at of inter tonib. The fitted annual terms of these time-series are compared to the offlerent generators and compared to the conter of the series of the conter of the c
We quartified the GPS and ELR00016 whit contenting and how this impacts the wohit read at more over the globa, which is a samilated into mean saa level (MSL) error, from the omission of the annual term of the geocenter correction. We find that for the SLR00016 statistical and how this impacts the wohit read at more over the globa, which is a samilated into mean saa level (MSL) error, from the omission of the annual term of the geocenter correction. We find that for the SLR00016 statistical and how this impacts the wohit read at more over the globa, which is a samilated into mean saa level (MSL) error, from the omission of the annual term of the geocenter correction. We find that for the SLR00016 statistical and how this impacts the wohit read at more over the globa, united in a same saa level (MSL) error, from the orbit frame which creates 1.66 ± 2.56 mm of systematics error in the MSL estimates.
REFERENCES ADMANDANCES J. F. J. C. Lemins, N. P. Zdenski, D. D. Buchenk, S. J. Lande, O. Berkuger, 2012, The effect of successor metrics an Journal. 2 mbits and de meas and Journal for the Journal Science Journal (Journal Journal), J. Science J. J. Science J. Sci

Introduction

CM ha e stations are subject, is a coupled problem. For ex stem causes geocenter motion and as such season witt et al., 2001). Métivier et al. (2010) have found th st probably between 0.3 and 0.8 mm/yr. As such, for re it well independently (Wu et al., 2012). Furthermor ng to the pri masses (Dong e mm/yr a the CM and t al., 2003, ind is today most prob rough to measure it we the purpose of an e, the CM is direct ations, having access to a nearly instantaneous geocenter is ex of the CM w

From the above, and given the required sea level infrastructure stability of 0.1 mm/r (Carenave s coordinates of GPS, SLR, and DORIS stations. This movement can be thought of as a global titherto, the lack of a community consensus on a geocenter model has not allowed the good orbital frame for altimetry centered in the CM plays a major role in the definition and accurate the statement of th 2021 accenter motion of the CM with respect to the CP, ideally, should also be included in the process of precise orbit if ex-1 loading displacement correction to be applied to the viral. Next coordinates of the tracking network in order to refer to be forward modeled as part of the Jason altimetry orbit standards (Cerri et al. 2010). Therefore, our motivation for this labit on of the rates of oldoal MS. There. letermination (POD), which is based of rence them to the CM of the whole Ea obal dec

Table 1: Description of the Jason-2 orbit solutions used in this study								
POD name used in the text	Description							
gpsdyn	GSFC's GPS dynamic							
gpsred	GSFC's GPS reduced dynamic							
gpsdyn_com_csr_	GSFC's GPS dynamic + Cheng et al. 2010 CoM correction							
gpsdyn_com_swn_	GSFC's GPS dynamic + Swenson et al. 2008 CoM correction							
SLR/DORIS	GSFC's SLR/DORIS dynamic							
SLR/DORIS_com_csr	GSFC's SLR/DORIS dynamic + Cheng et al. 2010 CoM correction							
SLR/DORIS_com_swn	GSFC's SLR/DORIS dynamic + Swenson et al. 2008 CoM correction							
jpl11a	JPL's release-11a GPS reduced dynamic							
The partly (see	the containing of the 7 companyout							

et al. (2010) and Cheng et al. (2010) inside the ises removed) as a co ng to Dong et al. (2003

 $X^{CH}(t) = X^{CF}(t) + X^{CH}_{CF}(t)$ (D) Where



Fig. 2: ation between the gpsdyn (reference orbit) and the test orbits: SLR/DORIS, gpsred. a) compared to the Swenson et al. (2010) applied only in the SLR/ its, b) compared to the Cheng et al. (2010) applied only to the SLR/DORIS orbits jpl11a



Spurious signals in the Jason-2 orbit origin

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Fig. 1: Periodogram (in mm) of the orbit origins after a 7-parameter Helmert transformation between the NASA GSFC Jason-2 GPS-based dynamic orbits and the three test orbits: NASA GSFC Jason-2 GPS-based reduced-dynamic (*gsred*), NASA GSFC Jason-2 GPS-based dynamic and PL Jason-2 GPS-based reduced-dynamic (*glitla*), a X-component, b Y-component, 0 Z-component. In purple, blue and orange are the comparisons to GSFC's *gpsred*, *SLR/DORIS* dynamic orbits and *jpl11a* GPS-based reduced dynamic orbits respectively.

a) & b) 118-day signal is dominant in the X and Y components with the largest amplitudes of 2.8 mm and 2.3 mm respectively. The 118-day signal is the precise dracontic (beta-prime) period for the Jason satellites and this result supports the earlier discussion about the remaining orbit enror due to salar radiation pressure (SRP) mis-modeling by Cerri et al. (2010) and Zelensky et al. (2010). c) in the Z-component the annual signature has the largest amplitude.

Shift in the orbit frame				Amplitudes of the propagated signals in the Z-component					
Geocenter model	Annual Amplitude		Ratio of the	Ratio of					Ratio of the resulted
Swenson et al. 2010	1.85	Geocenter model applied	reduction to the SLR/DORIS – gpsdyn signal	the reduction to each model	Helmert transformation	Ref. Orbit	Amplitude (mm)	Phase (degrees)	cionature
Cheng et al. 2010	4.24								to each
Helmert transformation (ref.	Annual								model
orbit gpsdyn)	Ampinuae				gpsdyn_com_swn	gpsdyn	0.3	62.7	16 %
SLRDORIS	2.82				gpsdyn_com_csr	gpsdyn	0.8	4.9	19 %
SLR/DORIS_com_swn	2.10	al. 2010	25 %	39 %	SLR/DORIS com csr	SLR/DORIS	3.1	7.2	74 %
SLR/DORIS com est	1.17	Cheng et al. 2010	58 %	39 %	SLR/DORIS_com_swn	SLR/DORIS	1.5	64.3	81 %

 Table 2:
 Z-component annual amplitudes (mm) from each geocenter motion model and orbit transformations compared to the ratios of reduction in the annual signature to each geocenter motion model and the SLR/DORIS – gpsdyn comparison
Table 3: Z-component annual amplitudes (mm) and ratios from each orbit solution after the geocenter motion correction to each model

Geocenter motion and mean sea level Phases of the geocenter motion correction as this propagates over the globe into the orbit's radial component Geographical distribution of the amplitudes of the radial orbit differences 200 200 200 100 210 Fig. 3 Amplitude (in mm) of the geocenter motion correction as it maps into the radial orbit differences (DH) of the gosdyn (up) and the SLR/DORIS (bottom) orbit frame. Left from Cheng et al. (2010) and right from Swenson et al. (2010). Fig. 4 Phase (in degrees) of the geocenter motion corrections as it maps into the radial orbit differences (DH) of the gpsdyn (up) and the SLR/DORIS (bottom) orbit frame. Left from Cheng et al. (2010) and right from Swenson et al. (2010). motion affects more the SLR/DORIS orbits. n in the GPS and SLR/DORIS orbits. The geographical distribution of the MSL from the CoM omission error Propag Orbit comparisons Ref. Orbit Tz (mm) DH (mm) DH/Tz SLR/DORIS_com_csr SLR/DORIS -4.67 ± 3.40 1.06 ± 2.66 -0.22 120 gpsdyn_com_csr gpsdyn -0.82 ± 0.28 0.17 ± 0.37 -0.21 onclusions/Future work



Fig. 5 Observed geographical MSL error (in mm) resulting from the geocenter motion model of the SLR/DORIS stations from Cheng et al. (2010) for Jason-2 cycle 058 (Jan 28-Feb 07, 2010)